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Reliable Calculations of Heat and Fluid Flow during Conduction Mode Laser Welding through Optimization of Uncertain Parameters

A deterministic approach is proposed to improve reliability of heat transfer and fluid flow calculations

BY A. DE AND T. DebROY

ABSTRACT. During conduction mode laser beam welding, the quality of numerical simulation of heat transfer and fluid flow in the weld pool is significantly affected by the uncertainty in the values of absorptivity, effective thermal conductivity, and effective viscosity that cannot be easily prescribed from fundamental principles. Traditionally, values of these parameters are either prescribed based on experience or adjusted by trial and error. This paper proposes a deterministic approach to improve reliability of heat transfer and fluid flow calculations. The approach involves evaluation of the optimized values of absorptivity, effective thermal conductivity, and effective viscosity during conduction mode laser beam welding from a limited volume of experimental data utilizing an iterative multivariable optimization scheme and a numerical heat transfer and fluid flow model. The optimization technique minimizes the error between the predicted and the measured weld dimensions by considering the sensitivity of weld dimensions with respect to absorptivity, effective thermal conductivity, and effective viscosity. Five sets of measured weld pool dimensions corresponding to five different welding conditions were utilized for the optimization. However, the procedure could identify the optimized values of the three uncertain parameters even with only three sets of measured weld pool dimensions.

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Introduction

Since the temperature and velocity fields in the weld pool are difficult to measure experimentally (Refs. 1-7), these important variables are often estimated by numerically solving the equations of conservation of mass, momentum, and energy. In recent years, the numerically computed temperature fields have been utilized to estimate weld pool dimensions (Refs. 4–7) and understand weld metal phase composition (Refs. 8-11), grain structure (Refs. 10, 11), inclusion structure (Refs. 12-14), and weld metal composition changes owing to both vaporization of alloying elements (Refs. 15, 16) and dissolution of gases (Refs. 17, 18).

The transport phenomena-based numerical models have been continually updated to include more detailed and realistic descriptions of component physical processes for simple(Refs. 19–22) as well as for complex weld joint geometries (Ref. 23). In recent years, these models have become relatively easy to use because of advances in computational hardware and software. However, these powerful numerical heat transfer and fluid flow models have not found widespread use in man-

KEY WORDS

Laser Laser Beam Welding Heat Transfer Fluid Flow Conduction Mode ufacturing or design applications. An important difficulty is the uncertainty involved in specifying some of the necessary input variables such as absorptivity, effective thermal conductivity, and effective viscosity. Although the time-tested physical laws such as the equations of conservation of mass, momentum, and energy provide a reliable phenomenological framework for calculations, the reliability of the numerical process models greatly depends on the accuracy of several input parameters.

Many input parameters necessary for the numerical simulation of heat transfer and fluid flow in conduction-mode linear laser beam welding can be readily specified. These include welding speed, beam power, beam diameter, and thermophysical properties of the material being welded (Refs. 19, 24). However, the values of absorptivity, effective thermal conductivity and effective viscosity cannot be specified from fundamental principles (Refs. 2, 24–30). For example, absorptivity depends on the chemical composition of the substrate, the surface finish, laser mode, and the prevailing temperature distribution on the weld pool. As a result, the absorption coefficient cannot be estimated theoretically with high reliability. However, an accurate value of absorptivity is critical for the dependable estimation of the rate of heat absorption. Similarly, appropriate values of effective thermal conductivity and effective viscosity are needed for the reliable modeling of the high rates of transport of heat, mass, and momentum in weld pools with strong fluctuating velocities (Ref. 25). Enhanced values of liquid thermal conductivity and viscosity have been frequently used to take into account the effects of the fluctuating



Fig. 1 — Influence of k^* and μ^* on (i) p_m^* and (ii) w_m^* with assumed absorptivity (η) of 0.30. Welding parameters: P = 3200 W, v = 3.33 mm/s ($N_{HI} = 21.90$).

Iuoie I	incusureu vi	era Dimensions,	incluing I ulu	meters (nen	2 i), und meut inpu	l Index
Data	Laser	Weld	Spot	N_{HI}	Weld	Weld
Set	Power	Velocity	Radius		Penetration	Width
Index	(W)	(mm.s ⁻¹)	(mm)		(mm)	(mm)
1	3500	8.33	1.3	9.67	1.00	4.00
2	5000	8.33	1.3	14.97	1.25	5.25
3	3200	3.33	1.4	21.90	1.75	4.00
4	4800	3.33	1.4	32.90	2.50	6.00
5	5000	3.33	1.3	34.53	2.25	6.75

Table 1 — Measured Weld Dimensions Welding Parameters (Ref. 24) and Heat Input Index

Table 2 — Chemical Composition (wt-%) of	
High-Speed Steel Used for Welding	
Experiments ^(a)	

С	Cr	W	Мо	V	Со	Mn
0.92	3.88	6.08	4.9	1.73	0	0.26
Si	S	Ni	Р	Cu	Al	Fe
0.23	0.001	0.24	0.024	0.20	0.019	Bal.

(a) for data set index 1, 2, and 5 (Ref. 24).

components of velocities in the weld pool. In some cases, the two-equation k-ɛ turbulence model has also been used in estimating the effective viscosity and effective thermal conductivity in the weld pool (Refs. 26–28). However, the two-equation k-ɛ turbulence model contains several empirical constants that were originally estimated from parabolic fluid flow data in large systems. As a result, its applicability for the recirculating flow in small scale systems has not been adequately tested. Since the effective thermal conductivity and viscosity depend on the turbulent kinetic energy and other properties of convection, these parameters are system properties (Refs. 1, 2, 24–30) and their values depend on welding conditions, particularly the heat input.

The values of effective viscosity and thermal conductivity have been determined in this work as a function of heat input from a limited volume of measured weld pool dimensions for conduction mode linear laser beam welding (Ref. 24) utilizing an optimization algorithm and a numerical heat transfer and fluid flow model. In contrast with the effective viscosity or the effective thermal conductivity, the laser beam absorption coefficient is a materials property. Although it varies with temperature, the extent of the variation is normally much smaller than those of the effective thermal conductivity or the effective viscosity. It has been taken as a constant in this work for simplicity. The optimization algorithm minimizes the error between the predicted and the experimentally observed penetrations and the weld widths by considering the sensitivity of the computed weld pool dimensions with respect to the absorptivity, effective thermal conductivity, and effective viscosity. The sensitivity terms are calculated by running the heat transfer and fluid flow model several times for each measurement considering small changes in the absorptivity, effective thermal conductivity, and effective viscosity (Refs. 29, 30).

The approach determines the values of absorptivity, effective viscosity and thermal conductivity in an iterative manner starting from a set of their initial guessed values. In order to include the effects of laser power, spot diameter, and welding speed into one convenient variable during optimization, a nondimensional heat input variable, N_{HI} , is defined as

$$N_{HI} = \frac{\frac{P}{\pi_{rb}^2 \cdot v}}{\rho C_{PS}(T_L - T_a) + \rho L}$$
(1)

where P is the laser power (W), $r_{\rm b}$ the spot radius (m), v the welding velocity $(m \cdot s^{-1})$, C_{PS} the specific heat of the solid metal $(J \cdot kg^{-1} \cdot K^{-1})$, ρ the density $(kg \cdot m^{-3})$, L the latent heat of fusion $(J \cdot kg^{-1})$ and T_L and T_a are the liquidus and ambient temperatures (K), respectively. In Equation 1, the numerator represents the available laser power per unit volume and the denominator depicts the enthalpy required to heat a unit volume of metal from ambient temperature to liquidus temperature. The numerator in Equation 1 when multiplied by the absorptivity, η , provides the absorbed heat per unit volume. The optimization approach identifies a single value of absorptivity and a linear trend of effective thermal conductivity and effective viscosity with N_{HI} from a limited volume of measurements.

The work presented in this manuscript represents a significant improvement over the previous (Refs. 31–35) reverse model-



Fig. 2 — Influence of k^* and μ^* on O(f) with assumed absorptivity (η) of 0.30. Welding parameters: P = 3200 W, v = 3.33 mm/s ($N_{HI} = 21.90$).



The effect of volume of data on the outcome of the optimization was examined. First, the optimization was done using five sets of measured weld pool dimensions. Second, the optimization was also carried out with only three measured data sets of weld pool dimensions. The optimized values of the uncertain variables were almost identical in both cases.

Heat Transfer and Fluid Flow Simulation

Table 1 depicts five sets of measurements of weld dimensions and the corresponding welding parameters that have been used in the present investigation. The chemical compositions of the steels used are presented in Tables 2 and 3. The steel compositions conform to two different grades of high-speed steel (Ref. 24). The thermophysical properties of these

Table 3 — Chemical Composition (wt-%) of
High-Speed Steel Used for Welding	
Experiments ^(a)	

С	Cr	W	Мо	V	Со	Mn
0.21	0.21	< 0.05	0.05	< 0.02	< 0.05	1.52
Si	S	Ni	Р	Cu	Al	Fe
0.36	0.006	0.14	< 0.005	0.14	0.01	Bal.

(a) for data set index 3 and 4 (Ref. 24).

steels are given in Tables 4 and 5. The flow of liquid metal in the weld pool in a threedimensional cartesian coordinate system is represented by the following momentum conservation equation (Refs. 4, 21, 22, 36):

$$\rho \frac{\partial u_j}{\partial t} + \rho \frac{\partial (u_i u_j)}{\partial x_i} = \frac{\partial}{\partial x_i} \left(\mu \frac{\partial u_j}{\partial x_i} \right) + S_j$$
(2)

where ρ is the density, t is the time, x_i is the distance along the i = 1, 2 and 3 directions, u_j is the velocity component along the j direction, μ is the effective viscosity, and S_j is the source term for the jth momentum equation and is given as (Refs. 21, 22)

$$S_{j} = -\frac{\partial p}{\partial x_{j}} + \frac{\partial}{\partial x_{j}} \left(\mu \frac{\partial u_{j}}{\partial x_{j}} \right) - C \left(\frac{\left(1 - f_{L} \right)^{2}}{f_{L}^{3} + B} \right) u_{j} - \rho U \frac{\partial u_{i}}{\partial x_{i}} + Sb_{j}$$
(3)

where p is the pressure, f_L is the liquid fraction, B is a constant introduced to



Fig. 3 —Influence of k^* and μ^* on O(f) with assumed absorptivity (η) of 0.30. Welding parameters: P = 5000 W, v = 8.33 mm/s ($N_{HI} = 14.97$).

Table 4 — Data Used for Calculations of
Temperature and Velocity Fields(a)

Physical Property	Value
Liquidus temperature, T _I (K)	1700.0
Solidus temperature, $T_s(K)$	1480.0
Ambient temperature, T_{a} (K)	293.0
Density of liquid metal, ρ (kg/m ³)	$8.1 imes 10^3$
Thermal conductivity of solid,	25.08
k _s (W m ⁻¹ K ⁻¹)	
Thermal conductivity of liquid,	25.08
k_{I} (W m ⁻¹ K ⁻¹)	
Specific heat of solid,	711.0
C_{PS} (J kg ⁻¹ K ⁻¹)	
Specific heat of liquid,	711.0
C_{PI} (J kg ⁻¹ K ⁻¹)	
Temperature coefficient of	-0.5×10^{-3}
surface tension, dy/dT (N m-1 K-1)	1
Coefficient of thermal	1.5×10^{-6}
expansion, β (K ⁻¹)	
Viscosity of molten iron	6.7×10^{-3}
at 1823 K. u . (kg.m ⁻¹ s ⁻¹)	

(a) for data set index 1, 2, and 5 (Ref. 24).

avoid division by zero, C (= 1.6×10^4) is a constant that takes into account mushy zone morphology and S_{bj} represents both the electromagnetic and buoyancy source terms. The third term on the right-hand side (RHS) represents the frictional dissipation in the mushy zone according to the Carman-Kozeny equation for flow through a porous media (·Refs. 37, 38). The pressure field was obtained by solving the following continuity equation simultaneously with the momentum equation

$$\frac{\partial(\rho u_i)}{\partial x_i} = 0 \tag{4}$$

40.0



Fig. 4 — Progress of calculation with four sets of intitial guessed values using (i) LM method, (ii) CGPR method, and (iii) CGFR method. The initial guessed values are presented in Table 6.

Initial guess : set 1 Initial guess : set 2 10.0 Initial guess : set 3 - Initial guess : set 4 5.0 (ii) O(f 1.0 0.1 21 81 101 1 41 61 Number of iterations 44.0 Unknown parameters 38.0 32.0 26.0 20.0 14.0 8.0 2.0 14 18 22 26 30 34 6 10 38 Non-dimensional heat input, N_{μ}

Fig. 5 — Estimated optimum values of k^* and μ^* for all values of N_{HI} . (Optimum value of $\eta = 0.25$ for all values of N_{HI})

The total enthalpy H is represented by a sum of sensible heat h and latent heat content Δ H, i.e., H = h + Δ H where h = =C_p dT, C_p is the specific heat, T is the temperature, Δ H = f_LL, L is the latent heat of fusion and the liquid fraction f_L is assumed to vary linearly with temperature in the mushy zone ·(Ref. 4).

$$f_{L} = \begin{cases} \frac{1}{0} \frac{T - T_{S}}{T_{L} - T_{S}} & T_{s} \stackrel{T > T_{L}}{\leq} T_{L} \\ & T < T_{S} \end{cases}$$
(5)

where T_L and T_S are the liquidus and solidus temperature, respectively. The thermal energy transport in the weld workpiece can be expressed by the following modified energy equation (Refs. 4, 21):

$$\rho \frac{\partial h}{\partial t} + \rho \frac{\partial (u_i h)}{\partial x_i} = \frac{\partial}{\partial x_i} \left(\frac{k}{C_p} \frac{\partial h}{\partial x_i} \right)$$
$$-\rho \frac{\partial \Delta H}{\partial t} - \rho \frac{\partial (u_i \Delta H)}{\partial x_i} - \rho U \frac{\partial h}{\partial x_i}$$
$$-\rho U \frac{\partial \Delta H}{\partial x_i} \tag{6}$$

where k is the thermal conductivity. The effective thermal conductivity in the liquid weld pool is also a property of the specific welding system and not a fundamental property of the liquid metal. Therefore, the value of the effective thermal conductivity is not known. Since the weld is symmetrical about the weld centerline only half of the workpiece is considered. The weld top surface is assumed to be flat. The velocity boundary condition is given as (Ref. 4)

$$\mu \frac{\partial u}{\partial z} = f_L \frac{d\gamma}{dT} \frac{\partial T}{\partial x}$$
$$\mu \frac{\partial v}{\partial z} = f_L \frac{d\gamma}{dT} \frac{\partial T}{\partial y}$$
$$w = 0 \tag{7}$$

where u, v, and w are the velocity components along the x, y, and z directions, re-





Fig. 7 — Experimentally determined and computed weld pool geometry. The length of the black arrow shows the magnitude of the velocities and the solid lines show the isotherms. Welding parameters: P = 5000 W, $\eta = 0.25$, and v = 8.33 mm/s ($N_{HI} = 34.53$).

Fig. 6 — Computed values of p_m^* and w_m^* using the optimized set of k^* and μ^* for all values of N_{HF} .

spectively, γ is the surface tension, and T is the temperature. The w velocity is zero, since the liquid metal is not transported across the weld pool top surface. The heat flux at the top surface is given as

$$k\frac{\partial T}{\partial z} = \frac{dP}{\pi r_b^2} \exp\left(-\frac{d\left(x^2 + y^2\right)}{r_b^2}\right)$$
$$-\left(T^4 - T_a^4\right) - h_c\left(T - T_a\right) \tag{8}$$

where r_b is the laser beam radius, d is the beam distribution factor, P is the laser beam power, η is the absorptivity, σ is the Stefan-Boltzmann constant, h_c is the heat transfer coefficient, and T_a is the ambient temperature. The first term on the RHS is the heat input from the heat source, defined by a Gaussian heat distribution. The second and third terms represent the heat loss by radiation and convection, respectively. The boundary conditions are defined as zero flux across the symmetric surface (i.e. at y = 0) as (Refs. 4, 21)

$$\frac{\partial u}{\partial y} = 0, v = 0, \ \frac{\partial w}{\partial y} = 0, \text{ and } \frac{\partial h}{\partial y} = 0$$
(9, 10)

At all other surfaces, temperatures are taken as ambient temperature and the velocities are set to zero.

Optimization Procedure

Both the Levenberg-Marquardt (LM) and the conjugate gradient (CG) methods have been described in the literature (Refs. 39–42) and only the special features of their application are described here. The optimization of the absorptivity, effective thermal conductivity and effective viscosity begins with the construction of an objective function that depicts the difference between the computed and the measured values of weld dimensions.

Levenberg-Marquart (LM) Method

In the LM method, the search for the optimized values follows the direction of the objective function gradient with step size modification by an adjustable damping parameter after each iteration. In the CG method, the direction of optimization is a conjunction of objective function gradient direction and the previous iteration direction (Refs. 39-42). The objective function, O(f) is defined as

$$O(f) = \sum_{m=1}^{m} \left[\frac{p_m^c - p_m^{obs}}{p_m^{obs}} \right] + \sum_{m=1}^{M} \left[\frac{w_m^c - w_m^{obs}}{w_m^{obs}} \right] = \sum_{m=1}^{M} \left[p_m^* - 1 \right]^2 + \sum_{m=1}^{M} \left[w_m^* - 1 \right]^2$$
(11)

where $p_{\tilde{m}}$ and $w_{\tilde{m}}$ are the penetration and the width of the weld pool computed by the numerical heat transfer and fluid flow model, respectively, $p_{\tilde{m}}^{obs}$ and $w_{\tilde{m}}^{os}$ are the corresponding measurements at similar welding conditions and $p_{\tilde{m}}^*$ and $w_{\tilde{m}}^*$ are nondimensional and indicate the extent of over or underprediction of penetration and weld width, respectively. In Equation 11, the subscript m refers to a specific weld in a series of M number of total welds and f corresponds to the given set of three unknown parameters in nondimensional forms as

$$\{f\} \equiv \{f_1 \ f_2 \ f_3\} \equiv$$
$$\{k^* \ \mu^* \ \eta\} \equiv \left\{\frac{k_{eff}}{k_s} \ \frac{\mu}{\mu_{fl}} \ \eta\right\}$$
(12)

where k_s , μ_{fl} , k_{eff} , μ , and η are thermal conductivity of solid material at room temperature, viscosity of molten iron at 1823 K, effective thermal conductivity, effective viscosity of liquid metal, and absorptivity, respectively. Assuming that O(f) is continuous and has a minimum value, the LM method tries to obtain the optimum values of f_1 , f_2 , and f_3 by minimizing O(f) with respect to them. In other words, Equation 11 is differentiated with respect to f_1 , f_2 and f_3 , and each derivative is made equal to zero as

where f_i represents k^* , μ^* or, η . The variables p_m^c and w_m^c in Equation 13 are obtained from the numerical heat transfer and fluid flow calculations for a certain set of f_1 , f_2 , and f_3 , i.e. k^* , μ^* , and η . The partial derivatives in Equation 13 are referred as sensitivity of the computed weld width and penetration with respect to the unknown parameters. The values of the sensitivity terms are numerically calculated. For example, the sensitivity of p_m^* with respect to f_1 is calculated as



$$\frac{\partial p_m^*}{\partial f_1} = p_m^* \begin{pmatrix} f_1 + \delta f_1, f_2, f_3, \\ \text{other known parameters} \end{pmatrix} - \frac{p_m^* \begin{pmatrix} f_1, f_2, f_3, \\ \text{other known parameters} \end{pmatrix}}{\delta f_1} \qquad (12)$$

where δf_1 is very small compared with f_1 . The solution of Equation 13 is achieved when both p_m^* and w_m^* becomes close to one. In other words, the calculated values of p_m^c and w_m^c should be close to the corresponding measured values of p_m^{ohs} and w_m^{ohs} for all M welds. Since f_1 , f_2 , and f_3 do not explicitly appear in Equation 13, this equation needs to be rearranged so that it can serve as a basis for an iterative scheme to evaluate the optimum values of f_1 , f_2 , and f_3 . The procedure is explained in Appendix 1. The final form of equations to be solved is

$$[S] \{ \Delta f^k \} = -\{ S^* \}$$
(15)
where,

$$\left\{f_i^{k+1}\right\} = \left\{f_i^k\right\} + \left\{\Delta f_i^k\right\} \text{ for } i = 1,3$$
(16)

and $\{f_i^{k+1}\}$ refers to the three unknown increments after $(k+1)^{\text{th}}$ iteration. Equation 15 provides the solution of the three unknown increments, $\{\Delta f_i^k\}$ corresponding to the three unknown parameters.

Conjugate Gradient (CG) Method

W

In the conjugate gradient technique, the unknown parameters are iteratively searched in the following sequence (Refs. 40–42):

$$f_i^{k+1} = f_i^k - \beta^k d_i^k$$
 for $i = 1,3$ (17)
where f_i^{k+1} represents the values of the
three unknowns after $(k+1)^{\text{th}}$ iteration.

indicates the directions of search at the end of kth iteration corresponding to the unknowns f_1 , f_2 , and f_3 , and β^k is the size of the search step. Both dk and β^k are calculated for every iteration or step. The variable β^k tends to adjust the extent of increment in unknown parameters between successive iterations and logically should assume a value that will facilitate the condition of objective function minimum. Thus, β^k is calculated by minimizing the residual objective function $O(f_i)^{k+1}$



Fig. 8 — Progress of calculation with four sets of initial guessed values using (i) LM, (ii) CGPR, and (iii) CGFR methods with three sets of measurements ($N_{\rm HI} = 14.97$, 32.90 and 34.53). The initial guessed values are presented in Table 6.

41

21

$$\frac{\partial O(f_i)^{k+1}}{\partial \beta^k} = 0 \; ; \; \beta^k \ge 0$$
(18)

0.10

0.05

The directions of search, d_1^k , d_2^k , and d_3^k , at the end of kth iteration are calculated as a linear conjugation of the corresponding directions of search at the end of $(k-1)^{th}$ iteration and the respective residual gradient of the objective function, O(f), after kth iteration as

$$d_1^k = \nabla O(f_1^k) + \gamma^k d_1^{k-1}$$
$$d_2^k = \nabla O(f_2^k) + \gamma^k d_2^{k-1}$$
$$d_3^k = \nabla O(f_3^k) + \gamma^k d_3^{k-1}$$
(19)

where γ^k is a conjugation coefficient at the end of kth iteration. The coefficient γ^k is

obtained either by Equation 20a using Polok-Ribier's (CGPR) modification or by Equation 20b using Fletcher-Reeve's modification (CGFR)

61

Number of iterations

81

101

$$\gamma^{k} = \frac{\sum_{i=1}^{3} \left\{ \nabla O(f_{i}^{k}) \right\}^{2}}{\sum_{i=1}^{3} \left\{ \nabla O(f_{i}^{k-1}) \right\}^{2}}$$

for $k = 1, 2, ...$ and $\gamma^{0} = 0$ (20a)
$$\gamma^{k} = \frac{\sum_{i=1}^{3} \left\{ \nabla O(f_{i}^{k}) \right\} \left\{ \nabla O(f_{i}^{k}) - \nabla O(f_{i}^{k-1}) \right\}}{\sum_{i=1}^{3} \left\{ \nabla O(f_{i}^{k-1}) \right\}^{2}}$$

for $k = 1, 2, ...$ and $\gamma^{0} = 0$ (20b)

Table 5 — Data Used for Calculations of Temperature and Velocity Fields^(a)

Physical Property	Value
Liquidus temperature, T _L (K)	1800.0
Solidus temperature, $T_{s}(K)$	1760.0
Ambient temperature, $T_a(K)$	293.0
Density of liquid metal, ρ (kg/m ³)	7.2×10^{3}
Thermal conductivity of solid,	25.08
k _s (W m ⁻¹ K ⁻¹)	
Thermal conductivity of liquid,	25.08
$k_{L} (W m^{-1} K^{-1})$	
Specific heat of solid,	754.0
C_{PS} (J kg ⁻¹ K ⁻¹)	
Specific heat of liquid,	754.0
C_{PL} (J kg ⁻¹ K ⁻¹)	
Temperature coefficient of	-0.5×10^{-3}
surface tension, dy/dT (N m-1 K-1)	
Coefficient of thermal	1.5×10^{-6}
expansion, β (K ⁻¹)	
Viscosity of molten iron	6.7×10^{-3}
at 1823 K, µ _{fl} (kg.m ⁻¹ s ⁻¹)	

(a) for data set index 3 and 4 (Ref. 24).

Further details of these two approaches are given in Appendix 2. In the CG methods, the direction of search is important, since the solution may diverge if the direction of search loses sight of the optimal solution. In the LM method, a manual damping factor is used that continually tracks the search step (or increment) so that the optimal solution cannot move away from the last computed minimum value of the objective function.

There are two main limitations in finding property data by the coupled heat and fluid flow and optimization procedure described in this manuscript. They are 1) the accuracy of measured depth and width and 2) how strongly the depth and the width vary with the uncertain parameter (sensitivity).

Results and Discussion

The sensitivity of the computed weld pool dimensions with respect to the effective thermal conductivity, effective viscosity, and absorptivity were determined by several heat transfer and fluid flow calculations. Figures 1A and B depict a number of isocontours of the dimensionless penetration, p_m^{*}, and the dimensionless width, w_m^* , as a function of k^* and μ^* for data set No. 3 in Table 1. It is observed from Fig. 1A that the dimensionless penetration, p_m^* , increases with k^* or μ^* . However, the dimensionless width, w_m^* , decreases with k^* or μ^* as shown in Fig. 1B. For k^{*} values above 7.0, both p_m^{*} and w_m^{*} become fairly insensitive to μ^* . Furthermore, both p_m^* and w^{*}_m approached a value of unity at high values of k^* and μ^* . When both p_m^* and w_m^* are 1, p_m^c equals p_m^{obs} and w_m^c equals w_m^{obs} and the calculated results agree with the correTable 6 — Sets of Initial Guesses for the Unknown parameters, C2 and C4

Set 1	Set 2	Set 3	Set 4	Optimized values
$C_2 = 0.5$	$C_2 = 1.0$	$C_2 = 1.5$	$C_2 = 2.0$	$\begin{array}{l} C_2 = 0.252 \\ C_4 = 1.115 \\ \eta = 0.250 \end{array}$
$C_4 = 0.5$	$C_4 = 1.0$	$C_2 = 1.5$	$C_4 = 2.0$	
$\eta = 0.1$	$\eta = 0.2$	$\eta = 0.3$	$\eta = 0.4$	

sponding measured values.

The effects of k^* and μ^* on the computed weld pool dimensions can be explained as follows:

The dimensionless weld penetration, p_m^{*}, increases with k^{*} since high values of thermal conductivity facilitate rapid heat transport in the downward direction. However, the higher thermal conductivity also reduces the surface temperature gradient and the radial convective heat transport and, consequently, decreases. Higher values of u* lowers radial convection and the convective heat flow resulting in both lower weld width and slightly higher peak temperature. The higher peak temperature enhances downward heat conduction and increases penetration. Furthermore, as k^{*} is progressively increased, conduction becomes the dominant mechanism of heat transfer and changes in μ^* do not significantly alter either the peak temperature or the convective heat transfer rate. Thus, the weld pool dimensions do not change significantly with µ* at high values of k* as observed in Fig. 1A and B.

It is quite apparent that in addition to the variation in k^* or μ^* , any change in the value of absorptivity will further influence the results presented in Fig. 1A and B. Although it has been reported that η depends on laser power (Ref. 43), the absorptivity is a material property, and its exact value depends on factors such as the surface temperature. An increase in the value of absorptivity implies an enhancement in the heat absorption rate that leads to higher peak temperature, greater temperature gradient and larger computed weld pool dimensions for a specific set of k^* and μ^* . In contrast, a decrease in the value of absorptivity leads to smaller values of computed weld pool dimensions for a specific set of k^* and μ^* . Such a behavior was also demonstrated in the case of a GTA weld pool (Ref. 29). To keep the problem tractable, a single optimized value of absorptivity (η) for all the welding conditions considered in the present work is assumed.

Figure 2 shows that high values of k^* and μ^* are necessary to achieve good agreement between the computed and the experimental weld pool geometry, i.e., low values of objective function for data set No. 3 in Table 1. In contrast, Fig. 3 indicates that low values of both k^* and μ^* are necessary to reduce the objective function for data set No. 2 in Table 1. These apparently contrasting results are achieved for welds with different heat input indexes (N_{HI}) of 21.90 and 14.97 for data set Nos. 3 and 2, respectively. The results in Figs. 2 and 3 are consistent with the fact that k^{*} and μ^* are not materials properties and their optimum values depend on N_{HI}. To account for the same in the procedure of optimization of k^{*} and μ^* in a simplified manner, the following linear relationships are assumed for simplicity:

$$k^* = C_1 + C_2 N_{HI}$$

$$\mu^* = C_3 + C_4 N_{HI}$$
(21)

where C_1 and C_3 are the minimum values of the effective conductivity and effective viscosity, respectively, and C_2 , and C_4 are constants. Since k^* and μ^* equal 1 at low values of N_{HI}, the values of both C_1 and C_3 are taken to be one. Thus, the optimization routine is used to estimate the values of C_2 and C_4 for each N_{HI}.

Results in Figs. 2 and 3 also indicate that several combinations of k^* and μ^* may result in low values of O(f) for a given N_{HI}. In order to seek optimum values for k^* and μ^* for a particular N_{HI}, an additional constraint is useful to achieve a physically realistic solution. Since k^* and μ^* are related by the turbulent Prandtl number, Pr_T, its value (= 0.9) provides a useful constraint. In other words, out of many possible solutions, the specific combination of k^* and μ^* nearest to the line corresponding to Pr_T = 0.9 will be chosen as the final solution. Pr_T is defined as

$$\Pr_T = \frac{\mu_T C_{PL}}{k_T} \tag{22}$$

where $\mu_{eff} = \mu_L + \mu_T$, $k_{eff} = k_L + k_T$ and μ_T , k_T are the turbulent viscosity and thermal conductivity, respectively, and μ_L and k_L are the viscosity and thermal conductivity of the liquid, respectively. Finally, Equation 12 is modified as

$${f} = {f_1 f_2 f_3} = {C_2 C_4 \eta}$$
 (23)

A set of initial values of C_2 , C_4 and η is necessary to start the optimization calculations by all three methods indicated in Appendixes 1 and 2. It is apparent from Figs, 1 through 3 that the expression $(\partial O(f)/\partial f_{i=1,3}$ does not conform to a continuous, convex, or concave-type function and the solution of Equation 13 involves multiple local minima in the objective function. To address this difficulty, the optimization calculations have been performed with a number of initial values in the ranges of 0.10 to 3.0, 0.10 to 3.0, and 0.10 to 0.50 for C_2 , C_4 , and η , respectively. Table 6 presents four such representative sets of initial values for which the progress of the optimization calculations are presented in Figures 4(i) to (iii). All measured weld pool dimensions indicated in Table 1 were used for the optimization. These figures show that the minimum attainable value of O(f) is affected by the choice of the initial values in both LM and CG methods. In the LM method, the minimum value of O(f) was obtained when the 1st and 2nd set of initial values (Table 6) were used (Fig. 4i) while the 3rd and 4th set of initial values resulted in values of O(f) as 0.219 and 0.837, respectively. In both the CGPR and CGFR methods, a minimum value of O(f) = 0.129 was achieved when the 2nd set of initial values (Table 6) were used (Figs. 4ii and 4iii). The value of O(f) could not be reduced below 0.129 with various other sets of initial guesses using all three optimization procedures. The final optimum solutions of C_2 , C_4 and η corresponding to O(f) = 0.129 were the same regardless of the initial guessed values of C_2 , C_4 and η . For example, a set of initial values of C2, C4 and η equal to 1.0, 1.0, and 0.3, respectively, yielded nearly similar values of O(f) and the optimized solution of C_2 , C_4 and η were the same as those obtained with the 2nd set of initial values (Table 6). Although a reduction in the value of O(f) from 0.166 in the LM method to 0.129 in the CG method appears to be small, it should be noted that O(f) depicts the square of the summation of the actual errors representing the discrepancy between pc and pobs, and wc and wobs. This discrepancy can be important even for small values of O(f) and achieving the smallest possible value of O(f) is important.

A closed-form relationship between the dependent variables such as the weld dimensions and the unknown parameters does not exist. As a result, the lowest value of O(f) depends on the initial guessed values of the unknown parameters. At the start of the iterations, the initial few gradients of the objective function with respect to the unknown parameters affect the results of the subsequent iterations and, ultimately, the optimum value of O(f). One way to resolve the difficulty resulting from the absence of a closed-form relation between the objective function and the unknown parameters is to per-

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form the optimization with a number of initial values so as to arrive at a best possible optimum solution. Furthermore, the optimization in the present work was computationally intensive, since the number of numerical heat transfer and fluid flow calculations for each optimization iteration was equal to the number of variables multiplied by the number of welds for sensitivity calculation plus the number of welds for error verification. However, an appropriate choice of the initial values could significantly reduce the volume of the computational work. Based on the results of optimization, Equation (21) can be rewritten as

 $k^* = 1.0 + 0.252 N_{HI}$

 $\mu^* = 1.0 + 1.115 \text{ N}_{\text{HI}}$ (24) The effectiveness of the optimization process is evident from the fact that all three optimization methods resulted in nearly similar sets of the optimum values of C₂, C₄, and η .

Figure 5 shows that both k* and μ^* increase significantly with N_{HI} consistent with Equation 24. This behavior is expected since higher heat input leads to more rapid transport of momentum and heat. The computed values of p_m^* and w_m^* using the optimized values of k* and μ^* are plotted in Fig. 6 for all values of N_{HI}. A fairly satisfactory agreement is obtained between the computed and measured weld dimensions. The slight discrepancy between the computed and the experimental values can be attributed, at least in part, to the experimental errors.

Typical computed temperature and velocity fields are shown in Fig. 7. The results show that the liquid metal is transported from the middle of the pool outward due to a negative temperature coefficient of surface tension. The features of the computed temperature and velocity fields are typical of the Marangoni convection dominated laser melted pools and have been discussed in the literature (Refs. 16, 19, 20, 24, 26, 27).

A test for the effectiveness of the proposed deterministic model is to check if the values of the uncertain parameters can be evaluated from a relatively small volume of experimental data. For this purpose, the values of C_2 , C_4 , and η were determined from the experimental data for three welds with $N_{HI} = 14.97$, 32.90, and 34.53 in Table 1. Figures 8(i) through (iii) show the variation of the objective function with number of iterations for four sets of initial guesses presented in Table 6 using all three optimization methods. It is observed that the LM method can provide an approximate minimum value of O(f) $\oplus 0.089$ when the 1st and 2nd set of initial values (Table 6) are used - Fig. 8i. Both CGPR and CGFR routines could reach

the minimum values of O(f) only when the 2nd set of initial values (Table 6) is used -Figs. 8ii and 8iii. The minimum value of O(f) could be reached within ten iterations using the LM routine and around 80 iterations in both CGPR and CGFR methods. The values of C_2 , C_4 , and η were found to be 0.252, 1.115, and 0.253, respectively, corresponding to the minimum value of O(f) following CG methods. Following the LM method, the optimized values of C_2 , C_4 , and η were found to be 0.265, 1.138, and 0.228, respectively, when only three sets of experimental data were used in the optimization process. The optimized values of these parameters did not change significantly when all five sets of experimental data were used for optimization.

The values of k^* and μ^* estimated in the present work are within the range of enhancement factors reported in the literature. For example, values in the range of 30 to 100 for both k^* and μ^* were estimated through trial and error to achieve good agreement between the computed and the measured weld dimensions (Ref. 27). When the k- ε turbulence model with a spatially variable effective viscosity was used, a maximum value of 16 for µ* was reported for a stationary GTA weld pool (Ref. 26). Although the value of η and the relationships between N_{HI} and k*, and μ * are valid for the specific conditions of welding considered here, a similar approach can be adopted for other welding conditions (Refs. 29, 30). Since η , k^{*} and μ^* are linked with p_m^* and w_m^* through the equations of conservation of mass, momentum and energy rather than through a straightforward polynomial function, local minima should be avoided by repeating the procedure with several sets of initial values. The intensive computational work needed to determine the uncertain parameters results in enhanced reliability of the numerical modeling of heat transfer and fluid flow in the weld pool.

Summary and Conclusions

Reliability of numerical heat transfer and fluid flow calculations in the weld pool can be significantly enhanced by determining the optimized values of effective thermal conductivity, effective viscosity, and absorptivity from a limited volume of measured weld dimensions. Three versions of gradient-based optimization techniques could produce low values of an objective function and determine the three aforementioned parameters. Although the values of these parameters were independent of the optimization process, the volume of the calculations needed and the manner in which the optimized values were obtained depended on both the opti-

mization method selected and the initial guessed values of the parameters. The values of effective thermal conductivity and effective viscosity were found to be much higher than their corresponding molecular values and also depended on heat input. Correlations are proposed to determine these parameters from welding conditions. The use of the optimized values of absorptivity, effective thermal conductivity and effective viscosity, determined from a limited volume of experimental data and the proposed model, resulted in good agreement between the computed and the experimentally determined fusion zone geometry without the need to adjust these parameters by trial and error.

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Appendix 1

In order to explain the basic concept of the LM method, a simplified system involving three unknown parameters, f_1 , f_2 , and f_3 , and one dependent variable, p^*_m measured under five welding conditions is considered first. Equation 13 can be written for f_1 , f_2 , and f_3 as:

$$\sum_{m=1}^{5} \left[\left(p_m^* - 1 \right) \frac{\partial p_m^*}{\partial f_1} \right] = 0;$$

$$\sum_{m=1}^{5} \left[\left(p_m^* - 1 \right) \frac{\partial p_m^*}{\partial f_2} \right] = 0;$$

$$\sum_{m=1}^{5} \left[\left(p_m^* - 1 \right) \frac{\partial p_m^*}{\partial f_3} \right] = 0$$

(A1, A2, A3)

The values of the three unknowns, f_1 , f_2 , and f_3 cannot be directly obtained from the above equations since they do not appear explicitly in these equations. The symbols f_1 , f_2 , and f_3 resemble k^* , μ^* , and η (absorptivity), respectively. So, the dependent variable p_m^* is expanded using the Taylor's series expansion to explicitly contain values of increments and, f_1 , f_2 , and f_3 . Considering two successive iterations of p_m^* and taking only the first order terms

$$\begin{pmatrix} p_m^* \end{pmatrix}^{k+1} = \begin{pmatrix} p_m^* \end{pmatrix}^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_1} \Delta f_1^k + \\ \frac{\partial \left(p_m^* \right)^k}{\partial f_2} \Delta f_2^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_3} \Delta f_3^k$$
(A4)

where Δf_2^k , Δf_2^k and Δf_3^k are three unknown increments corresponding to f_1 , f_2 , and f_3 as

$$f_1^{k+1} = f_1^k + \Delta f_1^k$$

$$f_2^{k+1} = f_2^k + \Delta f_2^k$$

$$f_3^{k+1} = f_3^k + \Delta f_3^k$$
(A5)

and f_1^{k+1} , f_2^{k+1} , and f_3^{k+1} correspond to the values of three unknowns after $(k+1)^{th}$ iteration. Except Δf_1^k , Δf_2^k , and Δf_3^k , all other terms on the right hand side of Equation A4 are considered to be known. To solve for Δf_1^k , Δf_2^k , and Δf_3^k , Equations A1, A2 and A3 are first rewritten replacing p_m^* by $(p_m^*)^{k+1}$ as

$$\begin{split} &\sum_{m=1}^{5} \left[\left(\left(p_{m}^{*} \right)^{k+1} - 1 \right) \frac{\left(\partial p_{m}^{*} \right)^{k+1}}{\partial f_{1}} \right] = 0; \\ &\sum_{m=1}^{5} \left[\left(\left(p_{m}^{*} \right)^{k+1} - 1 \right) \frac{\left(\partial p_{m}^{*} \right)^{k+1}}{\partial f_{2}} \right] = 0 \\ &\sum_{m=1}^{5} \left[\left(\left(p_{m}^{*} \right)^{k+1} - 1 \right) \frac{\left(\partial p_{m}^{*} \right)^{k+1}}{\partial f_{3}} \right] = 0 \end{split}$$

(A6, A7, A8)

However, p_m^* equals to p_m^c/p_m^{obs} , and although p_m^{obs} is a known measured value, p_m^c is to be computed using the numerical heat transfer and fluid flow calculation for

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a set of f_1 , f_2 , f_3 and other known parameters. So, $(p_m^*)^{k+1}$ that is the value of p_m^* after $(k+1)^{th}$ iteration is unknown since Δf_1^k , Δf_2^k , and Δf_3^k are unknown. Next, substituting right hand side of Equation A4 in the place of $(p_m^*)^{k+1}$, Equations A6, A7, and A8 are rewritten as:

$$\sum_{m=1}^{5} \begin{bmatrix} \left(\left(p_m^* \right)^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_1} \Delta f_1^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_2} \Delta f_2^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_3} \Delta f_3^k - 1 \end{bmatrix} = 0$$
$$\frac{\partial \left(\left(p_m^* \right)^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_1} \Delta f_1^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_2} \Delta f_2^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_3} \Delta f_3^k \end{bmatrix}}{\partial f_1} \end{bmatrix}$$
(A9)

$$\sum_{m=1}^{5} \left| \frac{\left(\left(p_m^* \right)^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_1} \Delta f_1^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_2} \Delta f_2^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_3} \Delta f_3^k - 1 \right)}{\partial \left(\left(p_{m*} \right)^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_1} \Delta f_1^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_2} \Delta f_2^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_3} \Delta f_3^k \right)}{\partial f_2} \right| = 0$$
(A10)

$$\sum_{m=1}^{5} \left[\frac{\left(p_m^* \right)^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_1} \Delta f_1^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_2} \Delta f_2^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_3} \Delta f_3^k - 1 \right)}{\partial \left(\left(p_m^* \right)^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_1} \Delta f_1^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_2} \Delta f_2^k + \frac{\partial \left(p_m^* \right)^k}{\partial f_3} \Delta f_3^k \right)}{\partial f_3} \right] = 0$$
(A11)

Neglecting higher order differentials e.g

$$\frac{\partial}{\partial f_1} \left(\frac{\partial \left(p_m^* \right)}{\partial f_1} \Delta f_1^k \right)_{\text{etc., Equations A9, A10}}$$

and A11 are further simplified as:

$$\sum_{m=1}^{5} \left[\left(\frac{\left(p_{m}^{*}\right)^{k} + \frac{\partial\left(p_{m}^{*}\right)^{k}}{\partial f_{1}} \Delta f_{1}^{k} + \frac{\partial\left(p_{m}^{*}\right)^{k}}{\partial f_{2}} \Delta f_{2}^{k} + \frac{\partial\left(p_{m}^{*}\right)^{k}}{\partial f_{3}} \Delta f_{3}^{k} - 1 \right) \frac{\partial\left(p_{m}^{*}\right)^{k}}{\partial f_{1}} \right] = 0$$
(A12)

$$\sum_{m=1}^{5} \left[\begin{pmatrix} \left(p_m^*\right)^k + \frac{\partial \left(p_m^*\right)^k}{\partial f_1} \Delta f_1^k + \\ \frac{\partial \left(p_m^*\right)^k}{\partial f_2} \Delta f_2^k + \\ \frac{\partial \left(p_m^*\right)^k}{\partial f_3} \Delta f_3^k - 1 \end{pmatrix} \frac{\partial \left(p_m^*\right)^k}{\partial f_2} \right] = 0$$

$$\sum_{m=1}^{5} \left[\left(\frac{\left(p_{m}^{*}\right)^{k}}{\left(p_{m}^{*}\right)^{k}} \frac{\partial \left(p_{m}^{*}\right)^{k}}{\partial f_{1}} \Delta f_{1}^{k} + \frac{\partial \left(p_{m}^{*}\right)^{k}}{\partial f_{2}} \Delta f_{2}^{k} + \frac{\partial \left(p_{m}^{*}\right)^{k}}{\partial f_{3}} \frac{\partial \left(p_{m}^{*}\right)^{k}}{\partial f_{3}} \right] = 0$$
(A14)

Equations A12, A13, and A14 are next rearranged as

$$\sum_{m=1}^{5} \left[\frac{\partial \left(p_{m}^{*} \right)^{k}}{\partial f_{1}} \frac{\partial \left(p_{m}^{*} \right)^{k}}{\partial f_{1}} \right] \Delta f_{1}^{k} + \\ \sum_{m=1}^{5} \left[\frac{\partial \left(p_{m}^{*} \right)^{k}}{\partial f_{1}} \frac{\partial \left(p_{m}^{*} \right)^{k}}{\partial f_{2}} \right] \Delta f_{2}^{k} + \\ \sum_{m=1}^{5} \left[\frac{\partial \left(p_{m}^{*} \right)^{k}}{\partial f_{1}} \frac{\partial \left(p_{m}^{*} \right)^{k}}{\partial f_{3}} \right] \Delta f_{3}^{k} = \\ - \sum_{m=1}^{5} \left[\frac{\partial \left(p_{m}^{*} \right)^{k}}{\partial f_{1}} \left(\left(p_{m}^{*} \right)^{k} - 1 \right) \right]$$
(A15)

Equations A15, A16, and A17 can be expressed in matrix form as $[S]{\Delta f^k} = -{S^*}$ (A18)

 $[S]{\Delta f^k} = -{S^*}$ (4) where

$$[S] = \begin{bmatrix} S_{11} S_{12} S_{13} \\ S_{21} S_{22} S_{23} \\ S_{31} S_{32} S_{33} \end{bmatrix} =$$

$$\begin{bmatrix} S = \begin{bmatrix} S_{11} S_{12} S_{13} \\ S_{21} S_{22} S_{23} \\ \frac{S_{11} S_{12} S_{23} \\ \frac{S_{11} S_{12} S_{23} \\ \frac{S_{11} S_{12} S_{23} \\ \frac{S_{11} S_{12} S_{13} \\ \frac{S_{11} S_{13} \\ \frac{S_{11$$

(

and
$$\left\{S^*\right\} = \begin{cases}S_1^p\\S_2^p\\S_3^p\end{cases} = \\ \begin{cases}\sum_{m=1}^5 \frac{\partial \left(p_m^*\right)^k}{\partial f_1} \left(\left(p_m^*\right)^k - 1\right)\\\sum_{m=1}^5 \frac{\partial \left(p_m^*\right)^k}{\partial f_2} \left(\left(p_m^*\right)^k - 1\right)\\\sum_{m=1}^5 \frac{\partial \left(p_m^*\right)^k}{\partial f_3} \left(\left(p_m^*\right)^k - 1\right)\\ \end{cases}$$

 $\left\{\Delta f^k\right\} = \begin{cases}\Delta f_1^k\\\Delta f_2^k\\\Delta f_3^k\end{cases}$

(A20, A21)

Thus, Equations A1, A2, and A3 are modified to equation A18 where the three unknown incremental terms Δf_k^t , Δf_k^k , and Δf_3^t are explicitly defined in terms of the known quantities. The solution of Δf_k^t , Δf_k^k , and Δf_3^t are used next to obtain f_1^{k+1} , f_2^{k+1} , and f_3^{k+1} (expression A5) that are employed to compute $(p_m^c)^{k+1}$ using the numerical heat transfer and fluid flow model. Next, $O(f)^{k+1}$ is calculated as

$$O(f)^{k+1} = \sum_{m=1}^{5} \left(\left(p_m^* \right)^{k+1} - 1 \right)^2$$
 (A22)

Values of f_1 , f_2 , and f_3 are assumed to reach optimum when the calculated value of $O(f)^{k+1}$ is smaller than a predefined small number. For the two dependent variables p_m^* and w_m^* , Equation A19 is modified as

$$[S] = \begin{bmatrix} S_{11} S_{12} S_{13} \\ S_{21} S_{22} S_{23} \\ S_{31} S_{32} S_{33} \end{bmatrix}$$
(A23)

where

$$S_{ij} = \sum_{m=1}^{5} \left(\frac{\frac{\partial \left(p_m^*\right)^k}{\partial f_i} \frac{\partial \left(p_m^*\right)^k}{\partial f_j}}{\frac{\partial \left(w_m^*\right)^k}{\partial f_i} \frac{\partial \left(w_m^*\right)^k}{\partial f_j}} \right) \text{ for } i, j = 1 \text{ to } 3$$
(A24)

Equation A20 will be modified as

$$\begin{cases} S^{*} \\ S^{*} \\ S^{*} \\ S^{pw} \\ S^$$

Equations A5 and A21 do not change since the number of unknown parameters remains three. Furthermore, the sensitivity terms such as

$$\frac{\partial \left(p_m^*\right)^k}{\partial f_{\cdot}} \text{ or } \frac{\partial \left(w_m^*\right)^k}{\partial f_{\cdot}}$$

 σf_i σf_i (for i = 1 to 3) in Equation A18 often tend to be very small as the values of the unknown parameters f_1, f_2 and f_3 move close to the optimum. As a result, the matrix [S] tends to become singular. To avoid numerical instability, Equation A18 is further modified following the LM method as

$$\left(\left[S\right] + \lambda I\right) \left\{ \Delta f^{k} \right\} = -\left\{S^{*}\right\}$$
(A26)

where λ is a scalar damping coefficient, usually about 0.001, and I is a diagonal matrix given by (Ref. 42)

$$I = \begin{bmatrix} S_{11} & 0 & 0 \\ 0 & S_{22} & 0 \\ 0 & 0 & S_{33} \end{bmatrix}$$
(A27)

The product λI in Equation A26 ensures that the left-hand term in Equation A26 will remain nonzero even if the determinant of the matrix [S] is zero. The value of λ is usually increased or decreased by a factor of ten as the value of the objective function in subsequent iterations increases or decreases. This, in effect, ensures the reduction or enhancement in step size as the solution respectively tends to diverge or converge. The algorithm of the complete procedure using the LM method can be presented as follows:

Step 1. Guess initial values (e.g., k^{th}) of unknown variables set, { f_i^k } for i=1, 3 from Equation 12.

Step 2. Choose initial value of damping factor (λ).

Step 3. Compute the value of the objective function, $O(f^k)$ from Equation 11.

Step 4. Solve for the set of unknown in-

crements $\{\Delta f_i^k\}$ for i = 1, 3 from Equation A26.

Step 5. Compute $\{f_i^{k+1}\}$ for i = 1, 3 from Equation A5.

Step 6. Compute $O(f^{k+1})$ from Equation 11.

Step 7. If $O(f^{k+1}) \ge O(f^k)$, set $\lambda = 10 \lambda$; reject $\{f^{k+1}\}$; go back to step 4.

Step 8. If $O(f^{k+1}) < O(f^k)$, set $\lambda = 0.1 \lambda$. Step 9. Exit if $O(f^{k+1}) - O(f^k)$ " ε_1 and

 $\{f_i^{k+1}\}-\{f_i^k\}$ " ε_2 ; or go back to step 4. ε_1 and ε_2 are two small, predefined numbers.

Appendix 2

Considering the objective function defined in Equation 11 with two dependent variables, p_m^* and w_m^* , and three unknown parameters, f_1 , f_2 , and f_3 , Equation 17 can be written as

$$f_1^{k+1} = f_1^k - \beta^k d_1^k$$

$$f_2^{k+1} = f_2^k - \beta^k d_2^k$$

$$f_3^{k+1} = f_3^k - \beta^k d_3^k$$

 $J_3 = J_3 - p \ a_3$ (A28) where f_{k+1}^{k+1} , f_{k+1}^{k} , d_1^k , d_2^k , and β^k confirm to their definitions presented previously. In Equation 18, as $O(f_i)^{k+1}$ contains $(p_m^*)^{k+1}$ and $(w_m^*)^{k+1}$ obtained from numerical heat transfer and fluid flow code using values of f_{k+1}^{k+1} , f_{2}^{k+1} , and f_{3}^{k+1} , and $O(f_i)^{k+1}$ depends on f_{k+1}^{k+1} , f_{2}^{k+1} , and f_{3}^{k+1} . Thus, replacing $(f_i)^{k+1}$ in Equation 18 by f_{3}^{k+1} . Thus, replacing $(f_i)^{k+1}$ and substituting the right-hand side of Equation A28 in place of them, Equation 18 can be rewritten as

$$\frac{\partial O \begin{pmatrix} f_1^k - \beta^k d_1^k, f_2^k - \\ \beta^k d_2^k, f_3^k - \beta^k d_3^k \end{pmatrix}}{\partial \beta^k} = 0; \ \beta^k \ge 0$$
(A29)

Considering two dependent variables, and $p^{\ast}_{\ m}$, $w^{\ast}_{\ m}$,

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 $O\left(f_1^k - \beta^k d_1^k, f_2^k - \beta^k d_2^k, f_3^k - \beta^k d_3^k\right) \text{ can be expressed further as}$

$$O\left(f_{1}^{k} - \beta^{k}d_{1}^{k}, f_{2}^{k} - \beta^{k}d_{2}^{k}, f_{3}^{k} - \beta^{k}d_{3}^{k}\right)$$

$$\equiv \sum_{m=1}^{5} \left[p_{m}^{*}\left(f_{1}^{k} - \beta^{k}d_{1}^{k}, f_{2}^{k} - \beta^{k}d_{2}^{k}, f_{3}^{k} - \beta^{k}d_{3}^{k}\right) - 1\right]$$

$$+ \sum_{m=1}^{5} \left[w_{m}^{*}\left(f_{1}^{k} - \beta^{k}d_{1}^{k}, f_{2}^{k} - \beta^{k}d_{2}^{k}, f_{3}^{k} - \beta^{k}d_{3}^{k}\right) - 1\right]$$

(A30)

Substituting Equation A30 in Equation A29 and using Taylor's expansion, Equation A29 can be substantially rearranged to give β^k as (Refs. 41, 42)

$$\beta^{k} = \frac{\left[\left[\left(p_{m}^{*} - 1 \right) \left(\frac{\partial p_{m}^{*}}{\partial f_{1}^{k}} d_{1}^{k} + \frac{\partial p_{m}^{*}}{\partial f_{2}^{k}} d_{2}^{k} \right) \right] + \left[\left[\left(w_{m}^{*} - 1 \right) \left(\frac{\partial w_{m}^{*}}{\partial f_{3}^{k}} d_{3}^{k} + \frac{\partial w_{m}^{*}}{\partial f_{2}^{k}} d_{3}^{k} \right) \right] \right] + \left[\left[\left(w_{m}^{*} - 1 \right) \left(\frac{\partial w_{m}^{*}}{\partial f_{1}^{k}} d_{1}^{k} + \frac{\partial w_{m}^{*}}{\partial f_{2}^{k}} d_{2}^{k} + \frac{\partial w_{m}^{*}}{\partial f_{3}^{k}} d_{3}^{k} \right) \right] \right] + \left[\beta^{k} = \frac{\left[\left(\frac{\partial p_{m}^{*}}{\partial f_{1}^{k}} + \frac{\partial w_{m}^{*}}{\partial f_{3}^{k}} d_{3}^{k} + \frac{\partial w_{m}^{*}}{\partial f_{2}^{k}} d_{3}^{k} + \frac{\partial w_{m}^{*}}{\partial f_{3}^{k}} d_{3}^$$

Equations 20a and 20b can respectively be rewritten as

$$\gamma^{k} = \frac{\sum_{i=1}^{3} \sum_{m=1}^{5} \left\{ \left(p_{m}^{*} - 1 \right) \frac{\partial p_{m}^{*}}{\partial f_{i}^{k}} + \left(w_{m}^{*} - 1 \right) \frac{\partial w_{m}^{*}}{\partial f_{i}^{k}} \right\}^{2}}{\sum_{i=1}^{3} \sum_{m=1}^{5} \left\{ \left(p_{m}^{*} - 1 \right) \frac{\partial p_{m}^{*}}{\partial f_{i}^{k-1}} + \left(w_{m}^{*} - 1 \right) \frac{\partial w_{m}^{*}}{\partial f_{i}^{k-1}} \right\}^{2}}$$
(A32)
$$\left\{ \frac{\sum_{i=1}^{3} \sum_{m=1}^{5} \left\{ \left(p_{m}^{*} - 1 \right) \frac{\partial p_{m}^{*}}{\partial f_{i}^{k}} + \left(w_{m}^{*} - 1 \right) \frac{\partial w_{m}^{*}}{\partial f_{i}^{k}} \right\} \right\}}{\left(\left(w_{m}^{*} - 1 \right) \left(\frac{\partial p_{m}^{*}}{\partial f_{i}^{k}} - \frac{\partial p_{m}^{*}}{\partial f_{i}^{k-1}} \right) + \right\}}{\left(\left(w_{m}^{*} - 1 \right) \left(\frac{\partial w_{m}^{*}}{\partial f_{i}^{k}} - \frac{\partial w_{m}^{*}}{\partial f_{i}^{k-1}} \right) \right) \right\}} \right]}$$
$$\gamma^{k} = \frac{\sum_{i=1}^{3} \sum_{m=1}^{5} \left\{ \left(p_{m}^{*} - 1 \right) \frac{\partial p_{m}^{*}}{\partial f_{i}^{k-1}} + \left(w_{m}^{*} - 1 \right) \frac{\partial w_{m}^{*}}{\partial f_{i}^{k-1}} \right\}^{2}}{\left(A33 \right)}$$

for k = 1, 2 and, $\gamma^0 = 0$. Apart from the calculation of conjugate coefficient, γ^* , both CGPR and CGFR methods are the same.

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